

In the Specification

Please substitute the following paragraphs for the corresponding paragraphs being amended:

[0002] In a gas turbine engine air is pressurized in a compressor and mixed with fuel for generating hot combustion gases in a combustor. The hot gases are discharged from the combustor into a high pressure turbine which extracts energy therefrom for powering the compressor.

[0020] The first stage intermediate nozzle may be formed of a suitable superalloy, such as the same nickel-based superalloy used for the high pressure turbine nozzles, with a corresponding oxidation resistant coating such as platinum-aluminide. These high strength nozzle vanes ~~having~~ have an associated maximum allowable metal temperature which is slightly below the temperature of the combustion gases in the intermediate power turbine.